

FAST SUBSONIC COMBUSTION AS A FREE-INTERFACE PROBLEM

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Abstract

The paper is concerned with the recently identified fast, yet subsonic, combustion waves occurring in obstacle-laden (e.g. porous) systems and driven not by thermal diffusivity but rather by the drag-induced diffusion of pressure. In the framework of a quasi-one-dimensional formulation where the impact of obstacles is accounted for through a frictional drag term, an asymptotic expression for the wave propagation velocity D is derived. The propagation velocity is controlled by the temperature (T_+) at the entrance to the reaction zone rather than at its exit (T_b) as occurs in deflagrative combustion. The evaluated $D(T_+)$ dependence allows description of the subsonic detonation in terms of a free-interface problem. The latter is found to be dynamically akin to the problem of gasless combustion known for its rich pattern-forming dynamics.

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1 Introduction

Premixed gas combustion is the combustion of gaseous reactants which are perfectly mixed prior to ignition. The most distinctive feature of premixed combustion is its ability to assume the form of a self-sustained reaction wave propagating subsonically or supersonically at a well-defined speed. Apart from their technological relevance, combustion waves constitute a truly fascinating dynamical system, displaying an amazingly rich variety of phenomena such as non-uniqueness of possible propagation regimes, their birth (ignition) and destruction (extinction), chaotic self-motion and fractal-like growth, various hysteretic transitions, etc.

One of the most effective practices in the theoretical exploration of combustion waves is their description in terms of a free-interface problem where the reaction zone is considered as infinitely thin compared to the other length-scales involved.

The current study is concerned with the formulation and analysis of a free-interface problem associated with the recently identified new mode of subsonic combustion arising in hydraulically resisted flows (e.g. porous beds), and where the combustion wave is driven by the drag-induced diffusion of pressure, rather than thermal diffusivity as occurs in conventional unconfined flames. In the simplest case (the so-called small-heat release approximation) the emerging free-interface problem is described by a single filtration equation (Sec. 6),

$$\Pi_\tau = \gamma \nabla^2 \Pi, \tag{1.1}$$

where Π is the reduced pressure that vanishes far ahead of the reactive interface $f = 0$, and approaches unity far behind; γ is the ratio of specific heats. At the interface the following matching conditions are held,

$$\gamma[\nabla \Pi \cdot \mathbf{n}] = -\Phi(\Pi) = -v_f, \quad [\Pi] = 0 \tag{1.2}$$

where $\Phi(\Pi)$ is the Arrhenius-type source intensity, $\mathbf{n} = \nabla f / |\nabla f|$ is the normal to the interface, and $v_f = -f_\tau / |\nabla f|$ its normal velocity. For all its seeming simplicity the system (1.1-2) is quite rich dynamically exhibiting both regular and complex behavior (Sec.2).

The paper is organized as follows. Section 2 yields a brief physical overview of combustion in hydraulically resisted flows, singling out the calculation of the equilibrium planar wave as a crucial step in the passage to the free-interface formulation. Sections 3,4 deal with the planar wave and evaluation of the associated propagation velocity. The latter involves a non-conventional eigen-value problem solved by an asymptotic separation of variables in an appropriate phase-plane. In Section 5 the relations obtained for the planar wave are extended over quasi-steady and quasi-planar configurations. As a result one ends up with a reduced model for pressure-driven combustion based on a free-interface formulation. To test its dynamical well-posedness the model is simulated numerically.

2 Physical overview

As is well known there are basically two mechanisms controlling propagation of combustion waves in gaseous mixtures : molecular transport and adiabatic compression. The first mechanism pertains to deflagrative combustion and the second to detonation. Normally the adiabatic compression is provided by the shock and the resulting combustion wave (detonation) spreads at a supersonic speed. In conventional detonation the shock and adiabatic compression are two inseparable ingredients. This coupling, however, is not inevitable. In hydrodynamically resisted flows, as those developing in porous beds, the burning velocity may fall significantly below its thermodynamic, Chapman-Jouguet (CJ) value, and under certain conditions the propagation may well become *subsonic* and therefore *shockless* [1-4]. The shockless propagation still involves pressure peaks and is sustained by adiabatic compression which is now spread by the drag induced diffusion of pressure. The transition from supersonic to subsonic propagation normally occurs in a gradual manner, e.g. by dilution of the mixture. In the near-sonic propagation the presence or absence of shocks is not crucial and there is no major distinction between supersonic and subsonic combustion. The subsonic mode emerges as a continuous extension of the supersonic one and hence may well be referred to as *subsonic detonation*, two words that one would have thought could never be connected! The idea of extending the concept of the detonation over the subsonic domain is not new and has already been discussed by Mitrofanov in his 1987-review on non-classical combustion waves [15]. In the concluding lines of the paper the author wrote, "It would be appropriate to extend the notion of detonation over a certain subsonic range of wave velocities ($D < a_0$) where there is a continuous in D passage to this domain with the preservation of pressure and density peaks within the front". It is interesting that the issue was raised by the experimentalists ten years ahead of the theoretical substantiation of the phenomenon. In [2, 4] employing the classical Zeldovich-Fanno [21] formulation, it was shown that the conventional and subsonic modes are actually two special cases of the general drag-affected detonation and in the velocity-drag plane belong to the same curve of dynamically feasible states (Fig. 1). The conventional CJ - detonation corresponds to the limit of low hydraulic resistance while the subsonic detonation to the limit of strong resistance.

In the subsonic detonation, due to the reaction rate strong temperature dependence, the bulk of the reaction is localized in a narrow sublayer of the overall wave structure controlled by the pressure diffusivity and the wave speed. This observation suggests the possibility of describing the pertinent dynamical system in terms of a free-interface problem, advantageous for theoretical explorations. To make such a formulation possible one needs a relation between the intensity of the localized reaction rate and the interface temperature. This, in turn, requires an analytical expression for the reaction wave speed relative to the background flow-field. It transpires, how-

ever, that the rational evaluation of this parameter cannot be carried out along the lines of the classical Zeldovich-Frank-Kamenetsky approach employed in the deflagration theory [22]. The point is that in subsonic detonation the temperature within the reaction zone undergoes an abrupt increase rather than being a slowly varying function as occurs in deflagrative combustion. As a result the traditional multiple-scale technique developed for analytical tackling of deflagrations appears unsuitable here. In the current paper the difficulty is resolved by the asymptotic separation of variables in the pressure-concentration phase-plane.

Curiously enough, in a certain parameter range, the emerging free-interface problem becomes functionally identical to the problem of gasless combustion (self-sustained high-temperature synthesis) known for its rich pattern forming dynamics, involving among other things, galloping and spinning structures as well as period-doubling cascades and chaos [9-13].

3 Model

To describe a reaction wave spreading through a hydraulically resisted flow the presence of obstacles will be accounted for by means of the velocity-dependent drag-force term added to the momentum equation. The current study is focused on the subsonic detonation only where $D < a_0$ (Fig. 1). Moreover, we shall consider the limit $D \ll a_0$, pertinent to strong hydraulic resistance. In this case the impact of inertial effects may be discarded and the system of governing equations may be written as,

$$(c_v \rho T)_t + \nabla(c_v \rho T \mathbf{u} + p \mathbf{u}) = QW, \quad \text{energy} \quad (3.1)$$

$$(\rho C)_t + \nabla(\rho C \mathbf{u}) = -W, \quad \text{concentration} \quad (3.2)$$

$$\rho_t + \nabla(\rho \mathbf{u}) = 0, \quad \text{continuity} \quad (3.3)$$

$$\nabla p = \mathbf{f}, \quad \text{momentum} \quad (3.4)$$

$$p = (c_p - c_v) \rho T, \quad \text{state.} \quad (3.5)$$

Here W is the reaction rate defined by a one-step first-order Arrhenius kinetics,

$$W = A \rho C \exp(-E/RT), \quad (3.6)$$

where E is a constant, specific to the reaction and called its activation energy, R is the universal gas constant, Q is the heat release, C is the concentration of the deficient reactant, A is the pre-exponential factor. Other notations are conventional. The drag-force \mathbf{f} is specified by the classical Darcy - Forchheimer equation [14,15],

$$\mathbf{f} = - \left(\frac{\phi \nu}{K} \right) \rho \mathbf{u} - \left(\frac{\phi^3 F}{\sqrt{K}} \right) \rho \mathbf{u} |\mathbf{u}| \quad (3.7)$$

Here ν is the kinematic viscosity of the fluid, ϕ is the porosity, i.e. the fraction of the bulk volume of porous bed occupied by void space. K is the permeability (conductivity) of the porous bed with respect to permeation by a fluid. This parameter is independent of the nature of the fluid but depends on the geometry of the medium. F is the Forchheimer coefficient controlling the impact of inertial effects ignored in the Darcy's equation pertinent to the creeping flow limit. For packed-sphere beds K and F are expressed in terms of the porosity ϕ and the particle diameter d_p by [8],

$$K = \frac{\phi^3 d_p^2}{\Gamma(1 - \phi)^2}, \quad F = \frac{\Sigma}{\Gamma^{1/2} \phi^{3/2}}, \quad (3.8)$$

where Γ, Σ are empirical constants. According to [8], $\Gamma = 150$ and $\Sigma = 1.75$.

To reduce the number of parameters involved the effects due to molecular transport and heat losses are discarded. Aside from the obvious advantages for physical analysis (conservation of enthalpy), the adiabatic limit may serve as quite a legitimate approximation away from the propagability threshold where the Reynolds analogy is violated and the momentum loss emerges as a dominating influence.

4 Travelling-wave solution

As mentioned in Section 2 the key point in converting the above formulation into a free-interface problem is the evaluation of the propagation velocity D of the associated travelling-wave solution.

In the frame of reference attached to the steady planar wave Eqs. (3.1)-(3.4) yield,

$$d[\rho(u - D)c_v T + pu]/dx = QW, \quad (4.1)$$

$$d[\rho(u - D)C]/dx = -W, \quad (4.2)$$

$$d[\rho(u - D)]/dx = 0, \quad (4.3)$$

$$dp/dx = f, \quad (4.4)$$

The wave is assumed to propagate through an initially quiescent homogeneous pre-mixture whose temperature, pressure, density, and deficient reactant concentration are regarded as prescribed. Hence the boundary conditions are,

$$\begin{aligned} T(\infty) &= T_0, & C(\infty) &= C_0, & \rho(\infty) &= \rho_0, \\ p(\infty) &= p_0 = (c_p - c_v)\rho_0 T_0, & u(\infty) &= 0. \end{aligned} \quad (4.5)$$

Far behind the wave, due to the flow deceleration and the reactant consumption,

$$u(-\infty) = 0, \quad C(-\infty) = 0, \quad (4.6)$$

and the global integration of the enthalpy equation, (Eq.4.1)+ Q (Eq.4.2),

$$d[\rho(u - D)(c_v T + QC) + pu]/dx = 0, \quad (4.7)$$

yields

$$T(-\infty) = T_0 + QC_0/c_v = T_b, \quad p(-\infty) = (c_p - c_v)\rho_0 T_b = p_b \quad (4.8)$$

Thus, due to the boundary condition (4.8) and the absence of heat losses, the final temperature and pressure of burnt gas appear to be identical to those reached in the constant volume adiabatic explosion.

Integration of (4.3) and (4.7) subject to the boundary conditions (4.5), (4.6), (4.8) allows the flow velocity, temperature and the specific volume, $v = 1/\rho$, to be expressed in terms of the pressure and concentration.

Specifically,

$$u = D[1 - (Tp_0/T_0p)], \quad (4.9)$$

$$v = c_p(1 - \gamma^{-1})(T/p), \quad \gamma = c_p/c_v, \quad (4.10)$$

$$T = \gamma^{-1}T_b + (1 - \gamma^{-1})T_0(p/p_0) - \gamma^{-1}(T_b - T_0)(C/C_0). \quad (4.11)$$

At high activation energies ($E/RT_b \gg 1$) the reaction zone shrinks, and the on length-scale larger than its width the reaction rate may be treated as a localized source,

$$W \simeq M\delta(x), \quad (4.12)$$

where M is the source intensity (to be found).

Given(4.12), Eqs.(4.1)-(4.4) imply the following jump conditions at the interface ($x = 0$)

$$[\rho(u - D)c_v T + pu] = MQ, \quad (4.13)$$

$$[\rho(u - D)C] = -M, \quad (4.14)$$

$$[\rho(u - D)] = 0, \quad (4.15)$$

$$[p] = 0, \quad (4.16)$$

Since

$$C(x > 0) = C_0 \quad \text{and} \quad C(x < 0) = 0, \quad (4.17)$$

and by (4.3)(4.5) $\rho(u - D) = -\rho_0 D$, Eq. (4.14) readily implies,

$$M = C_0 \rho_0 D, \quad (4.18)$$

where at this stage D remains undetermined.

Outside the reaction zone the overall problem is easily tractable and one ends up with the following set of basic profiles.

At $x < 0$ (burnt gas),

$$C = 0, \quad T = T_b, \quad v = v_b = v_0, \quad u = 0, \quad p = p_b. \quad (4.19)$$

At $x \geq 0$ (fresh mixture),

$$\begin{aligned} C &= C_0, \quad T = \gamma^{-1} p_0^{-1} T_0 (p_0 + (\gamma - 1)p(x)), \\ v &= \gamma^{-2} (\gamma - 1) c_p p_0^{-1} T_0 (p_0 + (\gamma - 1)p(x)) / p(x), \\ u &= \gamma^{-1} D (p(x) - p_0) / p(x), \end{aligned} \quad (4.20)$$

where $p(x)$ is governed by the equation,

$$dp/dx = f(v(p), u(p)), \quad p(0) = p_0. \quad (4.21)$$

At the upstream (unburnt) side of the reaction zone,

$$T = T_+ = T_0 + (1 - \gamma^{-1})(T_b - T_0) \quad (4.22)$$

Thus within the reaction zone the temperature undergoes a sharp increase from T_+ to T_b (Fig. 2). As has been shown by Goldfarb *et.al* [11] T_+ plays a pivotal role in controlling the reaction time scale τ ,

$$\tau \sim \exp(E/RT_+), \quad (4.23)$$

not dissimilar to the situation with the post-shock (Neumann) temperature in supersonic detonation [21].

5 Evaluation of the propagation velocity

To evaluate the propagation velocity D , it is helpful to introduce a set of scaled quantities,

$$\begin{aligned} \Theta &= (T - T_0)/(T_b - T_0), \quad \Pi = (p - p_0)/(p_b - p_0), \quad \Psi = 1 - C/C_0, \quad V = v/v_0, \\ U &= u/D, \quad \xi = x/l, \quad \tau = Dt/l, \quad \sigma = T_0/T_b, \quad q = (1 - \sigma)/\sigma. \end{aligned} \quad (5.1)$$

Here the length scale l is defined by

$$l = qa_0^2 K^{1/2} / \gamma \phi^3 F D^2, \quad (5.2)$$

where $a_0 = \sqrt{\gamma(c_p - c_v)T_0}$ is the velocity of sound in the unburnt gas. The scaled version of Eqs.(4.2)(4.4) thus becomes,

$$\Lambda^3 \frac{d\Psi}{d\xi} = -\frac{1}{V}\Omega(\Psi, \Theta), \quad (5.3)$$

$$\frac{d\Pi}{d\xi} = -\frac{U(\kappa + |U|)}{V}, \quad (5.4)$$

where according to (4.9)-(4.11),

$$\Theta = (1 - \gamma^{-1})\Pi + \gamma^{-1}\Psi, \quad V = (1 + q\Theta)/(1 + q\Pi), \quad U = q(\Pi - \Psi)/\gamma(1 + q\Pi). \quad (5.5)$$

$$\Omega = (1 - \Psi) \exp [\beta(\Theta - (1 - \gamma^{-1})) / (\sigma + (1 - \sigma)\Theta)], \quad (5.6)$$

is the scaled reaction rate, where $\beta = (1 - \sigma)E/RT_+$ is the effective Zeldovich number. Λ is the scaled propagation velocity defined by,

$$\Lambda^3 = \gamma\phi^3 F D^3 \exp(E/RT_+) / A q a_0^2 K^{1/2}. \quad (5.7)$$

$$\kappa^{-1} = \phi^2 F D K^{1/2} / \nu, \quad (5.8)$$

is the effective Reynolds number.

Eqs.(5.3)(5.4) are considered jointly with the boundary conditions,

$$\Psi(\infty) = 0, \quad \Psi(-\infty) = 1, \quad \Pi(\infty) = 0, \quad \Pi(-\infty) = 1. \quad (5.9)$$

To circumvent the familiar cold-boundary-difficulty we shall introduce an auxiliary ignition temperature $\Theta = \Theta^*$ and redefine Ω so that,

$$\Omega = 0 \quad \text{at} \quad 0 < \Theta < \Theta^*. \quad (5.10)$$

For high enough β the final result should not depend on the specifics of the truncation. At $\Theta < \Theta^*$, due to the absence of diffusion, $\Psi = 0$, and hence by Eq.(5.5), $\Theta = (1 - \gamma^{-1})\Pi$. At the ignition point, therefore,

$$\Pi = \Pi^* = \Theta^*/(1 - \gamma^{-1}). \quad (5.11)$$

Assuming that $U > 0$ and hence $d\Pi/d\eta < 0$, the boundary conditions (5.9) immediately imply that $\Pi < 1$ and by (5.11), $\Theta^* < 1 - \gamma^{-1}$. Hence, Π may now be employed as a new independent variable reducing Eqs.(5.3)(5.4) to a single first-order O.D.E.,

$$\Lambda^3 \frac{d\Psi}{d\Pi} = -\frac{\Omega(\Psi, \Theta(\Psi, \Pi))}{U^2(\Psi, \Pi) + \kappa(\Lambda)U(\Psi, \Pi)}, \quad (5.12)$$

considered jointly with the boundary conditions,

$$\Psi(\Pi^*) = 0, \quad \Psi(1) = 1. \quad (5.13)$$

Being overdetermined the problem (5.12)(5.13) should produce both $\Psi(\Pi)$ and the scaled propagation velocity Λ .

For further analysis it is helpful to recast Eq. (5.12) as,

$$\Lambda^3 H(\Psi, \Pi) d\Psi = G(\Psi, \Pi) d\Pi, \quad (5.14)$$

where

$$G(\Psi, \Pi) = \gamma q^{-2} (1 + q\Pi)^2 \exp\left(\frac{\beta(1 - \gamma^{-1})(\Pi - 1)}{\sigma + (1 - \sigma)((1 - \gamma^{-1})\Pi + \gamma^{-1}\Psi)}\right),$$

$$H(\Psi, \Pi) = \frac{(\Pi - \Psi)(\Pi - \Psi + \gamma q^{-1}\kappa(\Lambda)(1 + q\Pi))}{\gamma(1 - \Psi)} \times \exp\left(-\frac{\beta\gamma^{-1}\Psi}{\sigma + (1 - \sigma)((1 - \gamma^{-1})\Pi + \gamma^{-1}\Psi)}\right). \quad (5.15)$$

Considering G as a function of Π , and H as a function of Ψ , Λ^3 is determined by the relation,

$$\Lambda^3 J = I, \quad (5.16)$$

where

$$I = \int_{\Pi^*}^1 G(\Psi(\Pi), \Pi) d\Pi \quad \text{and} \quad J = \int_0^1 H(\Psi, \Pi(\Psi)) d\Psi. \quad (5.17)$$

At large Zeldovich numbers Ψ is close to zero everywhere except for a small vicinity of $\Pi = 1$ where it sharply raises to unity. The asymptotic evaluation of the integrals (5.17) then yields (see Appendix),

$$I = \frac{\gamma(1 + q)^2(\sigma + (1 - \sigma)(1 - \gamma^{-1}))}{\beta q^2(1 - \gamma^{-1})} + O\left(\frac{\ln \beta}{\beta^{3/2}}\right), \quad (5.18)$$

$$J = \frac{(1 + \gamma q^{-1}(1 + q)\kappa(\Lambda))(\sigma + (1 - \sigma)(1 - \gamma^{-1}))}{\beta} + O\left(\frac{\ln \beta}{\beta^{3/2}}\right). \quad (5.19)$$

Substituting (5.18)(5.19) into (5.16) one readily obtains,

$$\Lambda^3 = \frac{\gamma(1 + q)^2}{q^2(1 - \gamma^{-1})(1 + \gamma q^{-1}(1 + q)\kappa(\Lambda))} + O\left(\frac{\ln \beta}{\sqrt{\beta}}\right). \quad (5.20)$$

To test the rate of convergence Figure 3 shows the numerical solution of the eigenvalue problem (5.12) (5.13) corresponding to $\kappa = 0$, $\Pi^* = 0.25$, $\sigma = 0.15$ ($q = 5.667$), $\gamma = 1.2, 1.3, 1.4$, and $5 < \beta < 50$. For example, at $\gamma = 1.3$, $T_0 = 295K^o$ and $E/RT_b = 15$, as follows from Eq.(4.22), $T_+ = 681K^o$ and hence $\beta = (1 - \sigma)E/RT_+ = 37$. In this case $\Lambda(37) = 1.95$, $\Lambda(\infty) = 1.98$ and thus $(\Lambda(\infty) - \Lambda(37))/\Lambda(\infty) = 0.015$, which is quite a small error.

Finally, upon returning to the original dimensional parameters, (5.20) yields,

$$\left(\frac{F\phi^3\sigma(1-\sigma)(\gamma-1)}{\gamma A a_0^2 \sqrt{K}} \right) D^3 + \left(\frac{\phi\nu\sigma(\gamma-1)}{A a_0^2 K} \right) D^2 = \exp\left(-\frac{E}{RT_+}\right). \quad (5.21)$$

At $A = 10^8 s^{-1}$, $E/R = 10,000K^o$, $T_0 = 295K^o$, $a_0 = 350m/s$, $d_p = 0.25cm$, $\nu = 0.15cm^2/s$, $\phi = 0.4$, $\sigma = T_0/T_b = 0.15$, $\gamma = 1.3$, Eqs.(4.22),(5.21) yield $T_+ = 681K^o$, $D = 74m/s$, $l = 23cm$ and $\kappa = 0.026$.

For fast reaction waves spreading through porous beds the parameter κ (5.8) is normally quite small which would justify omission of the linear (Darcy) term in Eq.(3.7) and, hence, the quadratic term in Eq.(5.21). As a result one ends up with a cubic-root dependence on the Arrhenius exponent,

$$D \sim \exp(-E/3RT_+). \quad (5.22)$$

Note that in the subsonic detonation the propagation speed is controlled by the temperature (T_+) at the entrance to the reaction zone rather than at its exit (T_b) as occurs in deflagrative combustion.

6 Free-interface problem and its numerical validation

The consumption rate, $C_0\rho_0D(T_+)$ (4.18), evaluated for steady planar waves will apparently hold also for the waves whose reaction zone is merely quasi-steady and quasi-planar. In this situation T_+ is no longer obliged to coincide with its thermodynamic value (4.22), but should be determined by the overall dynamics of the system. Due to the high activation energy even a mild variation in T_+ may strongly affect $D(T_+)$. This approach, long and successfully employed in deflagrative combustion, allows description of subsonic detonation in terms of a free-interface problem. The corresponding set of governing equations may thus be written as,

$$(c_v\rho T)_t + \nabla \cdot (c_v\rho T\mathbf{u} + p\mathbf{u}) = QC_0\rho_0D(T_+)\delta_F, \quad (6.1)$$

$$(\rho C)_t + \nabla \cdot (\rho C\mathbf{u}) = -C_0\rho_0D(T_+)\delta_F. \quad (6.2)$$

Here δ_F is the surface δ -function with the subscript F labeling the flame-interface, $F(\mathbf{x}, t) = 0$.

Eqs. (6.1)(6.2) should be considered jointly with Eqs. (3.3)-(3.5), (4.17)(5.21).

Due to (4.17), integration of Eq. (6.2) across the interface yields,

$$\rho_+(V_F - \mathbf{u}_+ \cdot \mathbf{n}) = \rho_0 D(T_+), \quad (6.3)$$

where $V_F = -F_t/|\nabla F|$ is the normal velocity of the interface, and $\mathbf{n} = \nabla F/|\nabla F|$ is its unit normal.

One thus ends up with a model involving only three equations (3.3)(3.4) and (6.1). Although the constructed free-interface formulation involves discontinuities, the drag induced pressure-diffusivity exerts a regularizing influence which should make the system well-posed dynamically, i.e. free of pathological instabilities as occurs, for example in the Darrieus-Landau [16], or Zaidel's [9] problems also dealing with combustion interfaces. To test this point the current model was simulated numerically for the simplest case of one-dimensional propagation (Figs. 4,5). As one might anticipate, at sufficiently low Zeldovich numbers (β) the equilibrium traveling-wave solution appears to be stable. Yet there is a critical value $\beta = \beta_{cr}(\sigma, \gamma)$ above which the travelling-wave loses its stability and the propagation occurs in an oscillatory (galloping) mode. Otherwise the model is quite well-behaved dynamically as far as pathological instabilities are concerned.

A more systematic approach to the stability question requires consideration of an eigen-value problem for a system of linear O.D.E-es with variable coefficients. The latter stem from the nonlinear convective terms and constitute a major obstacle for an analytical tackling of the problem. This difficulty, however, vanishes if one considers the limit of small-heat-release ($\sigma \rightarrow 1$) which, while suppressing convective transport, preserves much of the problem's character including its discontinuities. In this limit, as has been recently shown [3], the system of governing equations (3.1)-(3.7) reduces to a piecewise-linear problem for the pressure. In the appropriately chosen units (5.1) the resulting model reads,

$$\Pi_\tau = \gamma \nabla^2 \Pi + \Phi(\Pi) \delta_f, \quad (6.4)$$

$$v_f = \Phi(\Pi). \quad (6.5)$$

Here $v_f = -f_\tau/|\nabla f|$ is the normal velocity of the interface, $f(\xi, \tau) = 0$, and

$$\Phi(\Pi) = \exp\left(\frac{\hat{\beta}(\Pi - 1)}{2(\hat{\sigma} + (1 - \hat{\sigma})\Pi)}\right), \quad (6.6)$$

with

$$\hat{\beta} = \frac{(1 - \sigma)(1 - \gamma^{-1})(E/RT_b)}{1 - \gamma^{-1}(1 - \sigma)}, \quad \hat{\sigma} = \frac{\gamma\sigma}{\gamma - 1 + \sigma} \quad (\hat{\sigma} < 1) \quad (6.7)$$

Eq.(6.4) should be considered jointly with the condition, $\Pi \rightarrow 0$ far ahead of the advancing interface.

The reduced model, as one may notice, is formally identical to the free-interface model for gasless combustion (self-propagating high-temperature synthesis) if Π is regarded as the scaled temperature; $\hat{\beta}$, Zeldovich number and $\hat{\sigma}$, the unburnt/burnt phase temperature ratio ($\hat{\sigma} < 1$). In this context the model has been explored energetically in recent years and there is a sizeable volume of analytical and numerical results on its intrinsic dynamics. The latter, among other things, involves galloping and spinning waves as well as period-doubling cascades and chaos [9-12].

Note that for the reduced model (6.4)-(6.7) the basic traveling-wave solution becomes oscillatingly unstable as soon as the scaled Zeldovich number $\hat{\beta} \sim (1 - \gamma^{-1})(1 - \sigma)(E/RT_b)$ exceeds some critical value [3]. Thus, for all the distinction in the underlying physics, the instability criteria for both subsonic and supersonic detonation [20,21] appear to be quite similar in form.

7 Concluding remarks

The multiplicity of drag-affected detonation regimes shown on Figure 1 is qualitatively similar to that found by Stewart and Yao [19] in curved detonations. The major difference is that in the latter case the additional (curvature) source term appears in the conservation of mass equation instead of the momentum equation and the emerging detonation regimes appear to be essentially supersonic.

In general drag-force based models can represent fast reaction waves in hydraulically resisted flows only in some average sense. In the subsonic low Mach number detonation discussed in this paper the developing reaction wave width,

$$l_w = (p_b - p_0)(dp(+0)/dx)^{-1}, \quad (7.8)$$

is normally much larger than the porous bed particle diameter d_p and may easily extend over tens of centimeters [12]. For example, for the parameter set considered in Sec.4, $l_w = 74d_p = 18.5cm$. In such a situation the reaction wave is expected to be only mildly distorted by the porous bed granules, and the adopted formulation is likely to be valid not only qualitatively but also quantitatively.

Acknowledgments

These studies were supported in part by the U.S.-Israel Binational Science Foundation under Grant No. 98-00374, the Israel Science Foundation under Grants Nos. 67-01 and 574-00, and the European Community Program TMR-ERB FMRX CT18 0201.

A Appendix

In this section we construct an *a priori* estimate for the solution $\Psi(\Pi)$ of the problem (5.12)(5.13) and evaluate Λ at $\beta \rightarrow \infty$.

Step 1. Estimate of $\Psi(\Pi)$. Suppose that $\Lambda \sim O(1)$ then there exists a constant $\hat{M} \sim O(1)$ and constant $\check{M} \sim O(1)$ such that $\hat{M} > \Lambda > \check{M}$. Therefore $\hat{\kappa} > \kappa(\Lambda) > \check{\kappa}$, where $\hat{\kappa} = \kappa(\hat{M})$ and $\check{\kappa} = \kappa(\check{M})$. Then an upper solution of Eq. (5.12) with the second boundary condition (5.13) is defined as a solution of the equation,

$$\frac{d\tilde{\Psi}}{d\Pi} = \tilde{M}S(\tilde{\Psi}, \Pi), \quad (\text{A.1})$$

where $S(\Psi, \Pi)$ is any function satisfying the condition,

$$S(\Psi, \Pi) \leq G(\Psi, \Pi)/H(\Psi, \Pi) \quad \text{for any } \Pi \in (\Pi^*, 1) \text{ and } \Psi \geq 0 \quad (\text{A.2})$$

and \tilde{M} is any positive number such that $\tilde{M} \leq \hat{M}^{-3}$.

The solution $\tilde{\Psi}$ will obviously be as close to the true solution Ψ as G/H is close to S . Thus to have some idea about Ψ via $\tilde{\Psi}$ one has to ensure that S retains the basic qualitative features of G/H . Such a specific choice may be made if we observe that for any $\Pi \in (\Pi^*, 1)$ and $\Psi \geq 0$,

$$\frac{(1 - \Psi)(1 + q\Pi)^2}{(\Pi - \Psi)(\Pi - \Psi + \gamma q^{-1}\kappa(\Lambda)(1 + q\Pi))} \geq \frac{1}{1 + \gamma q^{-1}\hat{\kappa}(1 + q)}, \quad (\text{A.3})$$

$$\begin{aligned} \exp\left(\frac{\beta(1 - \gamma^{-1})(\Pi - 1)}{\sigma + (1 - \sigma)((1 - \gamma^{-1})\Pi + \gamma^{-1}\Psi)}\right) \\ \geq \exp\left(\frac{\beta(1 - \gamma^{-1})(\Pi - 1)}{\sigma + (1 - \sigma)(1 - \gamma^{-1})\Pi^*}\right), \end{aligned} \quad (\text{A.4})$$

$$\exp\left(\frac{\beta\gamma^{-1}\Psi}{\sigma + (1 - \sigma)((1 - \gamma^{-1})\Pi + \gamma^{-1}\Psi)}\right) \geq \exp(\beta\gamma^{-1}\Psi), \quad (\text{A.5})$$

and set

$$\begin{aligned} S = \frac{\gamma^2}{q^2(1 + \gamma q^{-1}(1 + q)\hat{\kappa})} \times \\ \exp\left(\frac{\beta(1 - \gamma^{-1})(\Pi - 1)}{\sigma + (1 - \sigma)(1 - \gamma^{-1})\Pi^*}\right) \exp(\beta\gamma^{-1}\Psi). \end{aligned} \quad (\text{A.6})$$

As for \tilde{M} we only assume that $\tilde{M} \sim O(1)$.

Separating variables in (A.1) with S defined by (A.6) and taking into account the boundary condition (5.13) at $\Pi = 1$, we have,

$$\begin{aligned} & \int_1^{\tilde{\Psi}} \exp(-\beta\gamma^{-1}s) ds \\ &= \frac{\gamma^2}{q^2(1+\gamma q^{-1}(1+q)\hat{\kappa})} \tilde{M} \int_1^{\Pi} \exp\left(\frac{\beta(1-\gamma^{-1})(s-1)}{\sigma+(1-\sigma)(1-\gamma^{-1})\Pi^*}\right) ds, \end{aligned} \quad (\text{A.7})$$

which gives

$$\begin{aligned} \tilde{\Psi}(\Pi) &= -\frac{1}{\gamma^{-1}\beta} \\ &\quad \times \ln\left(\tilde{M}_1 \left(1 - \exp\left(\frac{\beta(1-\gamma^{-1})(\Pi-1)}{\sigma+(1-\sigma)(1-\gamma^{-1})\Pi^*}\right)\right) + \exp(-\gamma^{-1}\beta)\right), \end{aligned} \quad (\text{A.8})$$

where $\tilde{M}_1 = \tilde{M}(\sigma+(1-\sigma)(1-\gamma^{-1})\Pi^*)/q^2(1+\gamma q^{-1}(1+q)\hat{\kappa})\gamma^{-1}(1-\gamma^{-1})$.

If we also require that

$\tilde{M} < q^2(1+\gamma q^{-1}(1+q)\hat{\kappa})\gamma^{-1}(1-\gamma^{-1})/(\sigma+(1-\sigma)(1-\gamma^{-1})\Pi^*)$ (which is always possible) $\tilde{\Psi}(\Pi)$ will then be positive for all $\Pi \in (\Pi^*, 1)$ and thus (A.8) is an upper solution (5.14) considered with both conditions (5.12).

The solution (A.8) possesses an interesting property (easily checked by direct substitution) that $\Psi \sim O(\beta^{-1})$ at $\Pi = 1 - \beta^{-1}$. Moreover $\Psi \sim O(\beta^{-1} \ln \beta)$ at $\Pi = 1 - \beta^{-n}$ where $n > 0$ is any fixed number. These observations imply that

$$\Psi(\Pi) \leq a_1\beta^{-1} \quad \text{for } \Pi \leq 1 - a_2\beta^{-1} \quad \text{and}, \quad (\text{A.9})$$

$$\Psi(\Pi) \leq a_3\beta^{-1} \ln \beta \quad \text{for } \Pi \leq 1 - a_4\beta^{-n}, \quad \text{for any } n > 0. \quad (\text{A.10})$$

Henceforth all constants a_i are assumed to be of order of unity.

It is also important to note that jointly with $\Psi(\Pi)$ we can also consider the inverse function $\Pi(\Psi)$ due to its monotonicity. The lower solution for this problem is merely an inversion of $\tilde{\Psi}(\Pi)$ which implies

$$1 - \Pi \leq a_5\beta^{-1} \quad \text{for } \Psi \geq a_6\beta^{-1} \quad \text{and}, \quad (\text{A.11})$$

$$1 - \Pi \leq a_7\beta^{-1} \ln \beta \quad \text{for } \Psi \geq a_8\beta^{-n} \quad \text{for any } n > 0. \quad (\text{A.12})$$

Step 2. Evaluation of Λ

To calculate Λ we have to obtain approximate values of the integrals I and J defined by (5.17).

Consider I first. Let us rewrite this integral as a sum of three terms,

$$\begin{aligned}
I &= I_1 + I_0 + I_{-1} = \\
&\int_{\Pi^*}^{1-1/\sqrt{\beta}} \gamma q^{-2} (1 + q\Pi)^2 \exp\left(\frac{\beta(1-\gamma^{-1})(\Pi-1)}{\sigma + (1-\sigma)((1-\gamma^{-1})\Pi + \gamma^{-1}\Psi(\Pi))}\right) d\Pi \\
&\quad + \int_{1-1/\sqrt{\beta}}^{1-1/\beta^2} \dots + \int_{1-1/\beta^2}^1 \dots \quad (\text{A.13})
\end{aligned}$$

The first and last terms in this expression are readily estimated as follows,

$$|I_{-1}| \leq a_9 \beta^{-2}, \quad |I_1| \leq a_{10} \exp(-a_{11} \sqrt{\beta}), \quad (\text{A.14})$$

while $I_0 \sim \beta^{-1}$. The main contribution to I therefore comes from I_0 . Let us calculate this integral. Note that by (A.10) $\Psi \leq a_3 \beta^{-1} \ln \beta$ for $\Pi \in (1 - \beta^{-1/2}, 1 - \beta^{-2})$ and therefore

$$\begin{aligned}
&\exp\left(\frac{\beta(1-\gamma^{-1})(\Pi-1)}{\sigma + (1-\sigma)((1-\gamma^{-1})\Pi + \gamma^{-1}\Psi(\Pi))}\right) \\
&= \left(1 + O\left(\frac{\ln \beta}{\sqrt{\beta}}\right)\right) \exp\left(\frac{\beta(1-\gamma^{-1})(\Pi-1)}{\sigma + (1-\sigma)(1-\gamma^{-1})\Pi}\right) \quad (\text{A.15})
\end{aligned}$$

which allows us to write,

$$I_0 = \tilde{I}_0 \left(1 + O\left(\frac{\ln \beta}{\sqrt{\beta}}\right)\right), \quad (\text{A.16})$$

where

$$\tilde{I}_0 = \gamma q^{-2} \int_{1-1/\sqrt{\beta}}^{1-1/\beta^2} (1 + q\Pi)^2 \exp\left(\frac{\beta(1-\gamma^{-1})(\Pi-1)}{\sigma + (1-\sigma)(1-\gamma^{-1})\Pi}\right) d\Pi. \quad (\text{A.17})$$

The last integral can be calculated using the Laplace technique which gives

$$\tilde{I}_0 = \frac{\gamma(1+q)^2(\sigma + (1-\sigma)(1-\gamma^{-1}))}{\beta q^2(1-\gamma^{-1})} + O\left(\frac{1}{\beta^2}\right). \quad (\text{A.18})$$

Thus,

$$I = \frac{\gamma(1+q)^2(\sigma + (1-\sigma)(1-\gamma^{-1}))}{\beta q^2(1-\gamma^{-1})} + O\left(\frac{\ln \beta}{\beta^{3/2}}\right). \quad (\text{A.19})$$

The second integral J can be calculated using the same procedure. First we rewrite it as a sum

$$J = J_{-1} + J_0 + J_1 = \int_0^{1/\beta^2} \gamma^{-1} \frac{(\Pi(\Psi) - \Psi)(\Pi - \Psi + \gamma q^{-1} \kappa(\Lambda)(1 + q\Pi))}{1 - \Psi} \times \exp\left(\frac{-\beta\gamma^{-1}\Psi}{\sigma + (1 - \sigma)((1 - \gamma^{-1})\Pi(\Psi) + \gamma^{-1}\Psi)}\right) d\Psi + \int_{1/\beta^2}^{1/\sqrt{\beta}} \dots + \int_{1/\sqrt{\beta}}^1 \dots \quad (\text{A.20})$$

The terms J_{-1} and J_1 are estimated as

$$|J_{-1}| \leq a_{12}\beta^{-2}, \quad |J_1| \leq a_{13} \exp(-a_{14}\sqrt{\beta}), \quad (\text{A.21})$$

and $J_0 \sim \beta^{-1}$ is the dominant term in J . Using (A.12) we conclude that for $\Psi \in (\beta^{-2}, \beta^{-1/2})$

$$\frac{(\Pi(\Psi) - \Psi)(\Pi - \Psi + \gamma q^{-1} \kappa(\Lambda)(1 + q\Pi))}{1 - \Psi} = 1 + \gamma q^{-1}(1 + q)\kappa(\Lambda) + O\left(\frac{\ln \beta}{\beta}\right), \quad (\text{A.22})$$

and

$$\exp\left(\frac{-\beta\gamma^{-1}\Psi}{\sigma + (1 - \sigma)((1 - \gamma^{-1})\Pi(\Psi) + \gamma^{-1}\Psi)}\right) = \left(1 + O\left(\frac{\ln \beta}{\sqrt{\beta}}\right)\right) \exp\left(\frac{-\beta\gamma^{-1}\Psi}{\sigma + (1 - \sigma)((1 - \gamma^{-1}) + \gamma^{-1}\Psi)}\right) \quad (\text{A.23})$$

Thus J_0 can be represented as

$$J_0 = \tilde{J}_0 \left(1 + O\left(\frac{\ln \beta}{\sqrt{\beta}}\right)\right), \quad (\text{A.24})$$

where

$$\tilde{J}_0 = \gamma^{-1}(1 + \gamma q^{-1}(1 + q)\kappa(\Lambda)) \times \int_{1/\beta^2}^{1/\sqrt{\beta}} \exp\left(\frac{-\beta\gamma^{-1}\Psi}{\sigma + (1 - \sigma)((1 - \gamma^{-1}) + \gamma^{-1}\Psi)}\right) d\Psi. \quad (\text{A.25})$$

Employing the Laplace technique we obtain

$$\tilde{J}_0 = \frac{(1 + \gamma q^{-1}(1 + q)\kappa(\Lambda))(\sigma + (1 - \sigma)(1 - \gamma^{-1}))}{\beta} + O\left(\frac{1}{\beta^2}\right), \quad (\text{A.26})$$

and hence

$$J = \frac{(1 + \gamma q^{-1}(1 + q)\kappa(\Lambda))(\sigma + (1 - \sigma)(1 - \gamma^{-1}))}{\beta} + O\left(\frac{\ln \beta}{\beta^{3/2}}\right). \quad (\text{A.27})$$

Substituting (A.19), (A.27) into (5.16) we finally obtain,

$$\Lambda^3 = \frac{\gamma(1 + q)^2}{q^2(1 - \gamma^{-1})(1 + \gamma q^{-1}(1 + q)\kappa(\Lambda))} + O\left(\frac{\ln \beta}{\sqrt{\beta}}\right). \quad (\text{A.28})$$

Since $\Lambda \sim O(1)$, our *a priori* assumption concerning its scaling is fully corroborated and therefore (A.28) yields a true asymptotics as $\beta \rightarrow \infty$.

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Figures

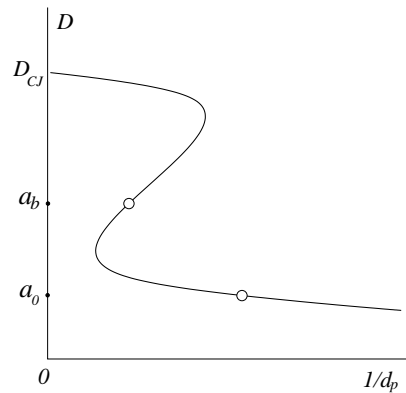


Figure 1.

Figure 1. Detonation velocity (D) vs. reciprocal of the particle diameter (d_p) [5,6]. D_{CJ} , a_0 , a_b correspond to the Chapman-Jouguet detonation and sound speeds in the fresh and burned gas, respectively.

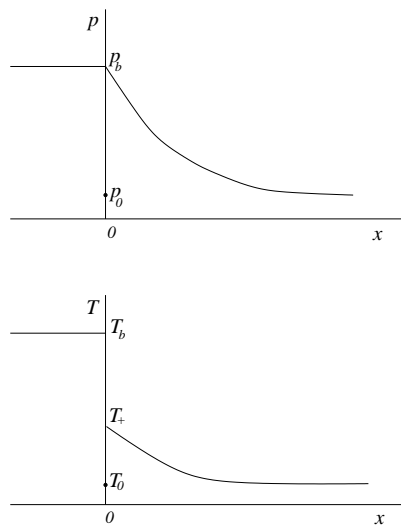


Figure 2.

Figure 2. Profiles of pressure (p) and temperature (T) in subsonic detonation (outer view).

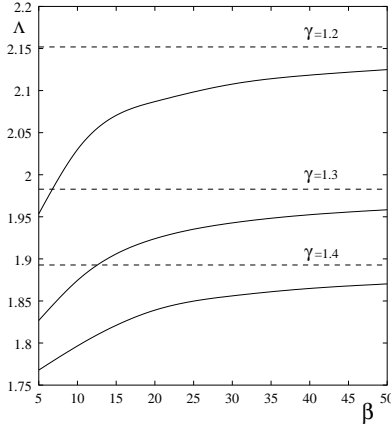


Figure 3.

Figure 3. Scaled propagation velocity (Λ) vs. scaled activation energy (β) evaluated for $\gamma = 1.2, 1.3, 1.4$, $\kappa = 0$ and $\sigma = 0.15$. Broken lines correspond to the asymptotics (5.20).

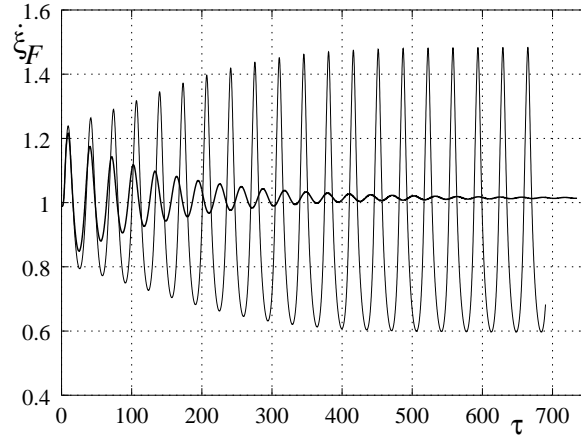


Figure 4

Figure 4. Scaled propagation velocity ($\dot{\xi}_F$) versus scaled time (τ) calculated for $\gamma = 1.3$, $\sigma = 0.15$, $\kappa = 0.025$, $\beta = 19$ and 20 . Bold and thin curves correspond to the intrinsically stable ($\beta = 19$) and galloping ($\beta = 20$) propagation.

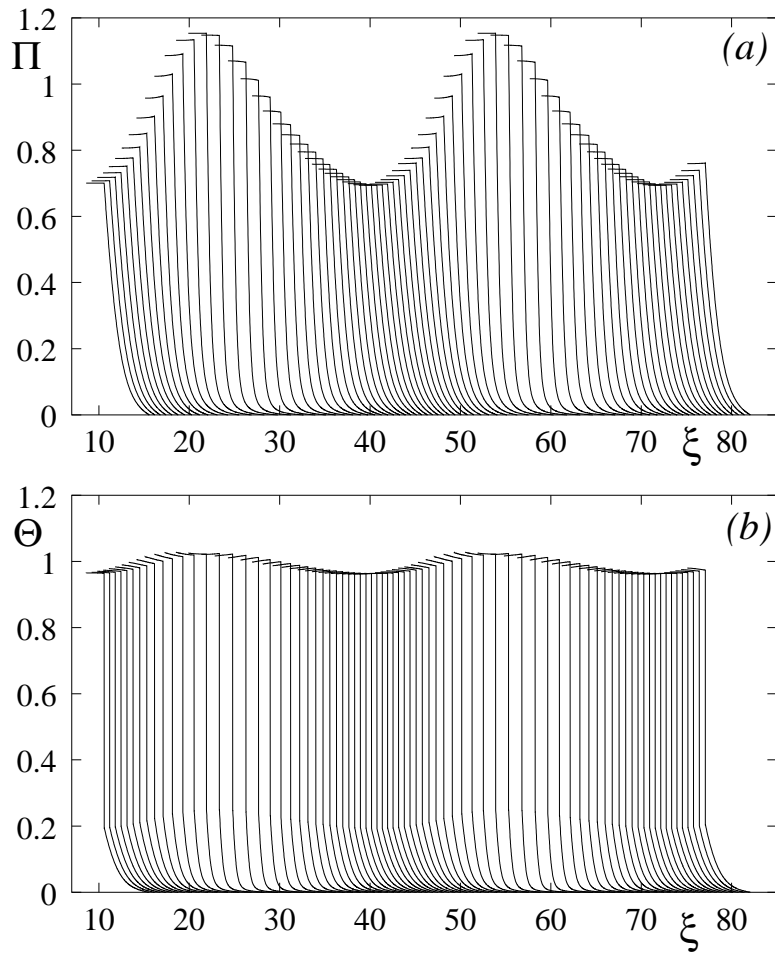


Figure 5

Figure 5. Galloping mode of subsonic detonation. Profiles of scaled pressure $\Pi(a)$ and temperature $\Theta(b)$ at several consecutive instants of time; $\gamma = 1.3$, $\sigma = 0.15$, $\kappa = 0.025$, $\beta = 20$.